3.1 Fluid Viscosity

For the Newtonian fluid, the dynamic viscosity μ [Pa \cdot s] is a fixed constant; whereas for the non-Newtonian fluid, the viscosity varies with the shear stress τ [Pa] and shear rate $\dot{\gamma}$ [1/s].

FIG. 1: Left: the concept of shear strain γ in a simple shear flow; Right: the rheological behaviour of viscous fluids can be classified by the shear stress - shear rate ($\dot{y} = dy/dt$) relations.

- Shear thickening: μ increases with shear rate $e.g.,$ ketchup;
- Shear thinning: μ decreases with shear rate $-e.g.,$ cornstarch paste;
- Bingham plastic: a yield stress τ_y impedes the fluid flow until $\tau>\tau_y.$

Although the blood is modelled as a Newtonian fluid, it is shear thinning with yield (*a.k.a.* Bingham pseudoplastic). The non-Newtonian behaviours of blood is due to the cell suspension (rather than the plasma), hence, the viscosity is Hematocrit-dependent.

3.2 Flow in a Rectangular Duct

Considerthe flow in a rectangular duct (length L, width w , height h) in the Cartesian coordinate system ([Figure 2](#page-1-0)).

Assumptions

- Fluid is homogeneous, incompressible and Newtonian with viscosity μ and density ρ ;
- Flow has reached the steady state: $\partial u/\partial t = 0$;
- Flow is fully developed along the x-direction: $\partial u/\partial x = 0$;
- Zero velocity along the ν and *z*-directions: $\nu = 0$, $\nu = 0$;
- Negligible body force: $f = 0$.

Boundary Conditions Symmetrical flow profile at $y = 0$ and $z = 0$; Non-slip condition at the wall $y = \pm h/2$, $z = \pm w/2$.

Aim Analytically solve for the flow velocity in the x -direction.

Solution The *x*-momentum equation is reduced to

$$
0 = -\frac{\partial p}{\partial x} + \mu \left(\frac{\partial^2 u}{\partial y^2} + \frac{\partial^2 u}{\partial z^2} \right).
$$

Using separation of variables^{[1](#page-0-0)}, the analytical solution of u is

$$
u = \frac{1}{2\mu} \frac{\partial p}{\partial x} \left[y^2 - \left(\frac{h}{2} \right)^2 - \sum_{n=0}^{\infty} A_n \cos \left(\frac{\lambda_n y}{h/2} \right) \cosh \left(\frac{\lambda_n z}{h/2} \right) \right], \quad \text{where } A_n = \frac{h^2 (-1)^n}{\lambda_n^3 \cosh \frac{\lambda_n w}{h}}, \quad \lambda_n = \frac{(2n+1)\pi}{2}.
$$

 $^{\rm 1}$ for the full derivation, see the Supplementary slides posted on Blackboard

FIG. 2: The schematic for the flow in a rectangular duct.

Integrating u over the area, the flux Q can be expressed as

$$
Q = \frac{\partial p}{\partial x} \frac{wh^3}{12\mu} \Big[6\Big(\frac{h}{w}\Big) \sum_{n=0}^{\infty} \lambda_n^{-5} \tanh\Big(\frac{\lambda_n w}{h}\Big) - 1 \Big] \approx \frac{\partial p}{\partial x} \frac{wh^3}{12\mu} \Big[1 - 0.6274\Big(\frac{h}{w}\Big) \Big].
$$

Finally, by $Q = \Delta p/R$, the flow resistance is

$$
R = \frac{\Delta p}{Q} = \frac{12\mu L}{wh^3 \left[1 - 0.6274 \left(\frac{h}{w}\right)\right]},
$$

3.3 Womersley Flow

Motivation To approximate the pulsatility nature of the flow in the cardiovascular system.

Assumptions

- Fluid is homogeneous, incompressible and Newtonian with viscosity μ and density ρ ;
- Flow in a long straight tube, with a perfect circular cross-section at radius a ;
- Axisymmetric along the θ -axis: $\frac{\partial}{\partial \theta} = 0$;
- The flow is fully developed along the *z*-axis: $\partial u/\partial z = 0$;
- No swirls: $u_{\theta} = 0$;
- No velocity along the radial direction: $u_r = 0$;
- Negligible body force: $f = 0$.

FIG. 3: The schematic of the Womersley flow in a pipe.

Boundary Conditions No-slip condition on the wall, parabolic condition as Poiseuille flow.

Solution Procedure

Step 1 The *z*-momentum equation

$$
\rho \left(\frac{\partial u_z}{\partial t} + u_y \frac{\partial u_z}{\partial r} + \frac{u_\theta}{\mathcal{A}} \frac{\partial u_z}{\partial \theta} + u_z \frac{\partial u_z}{\partial z} \right) = -\frac{\partial p}{\partial z} + \mu \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial u_z}{\partial r} \right) + \frac{1}{r^2} \frac{\partial^2 u_z}{\partial \theta^2} + \frac{\partial^2 u_z}{\partial z^2} \right] + \rho f_z^2
$$
\n
$$
\Rightarrow \rho \frac{\partial u_z}{\partial t} = -\frac{\partial p}{\partial z} + \mu \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial u_z}{\partial r} \right).
$$

Assume the pressure gradient is sinusoidal: $\partial p/\partial z = \frac{G_0}{2}$ $\frac{10}{2}e^{i\omega t}$, and following the sinusoidal *z*-velocity: $u_z =$ $U(r)e^{i\omega t}$:

$$
\left[i\omega U\rho + \frac{G_0}{2} - \mu \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial U}{\partial r}\right)\right] e^{i\omega t} = 0 \qquad \frac{2^{\text{nd}} - \text{order ODE}}{\frac{\partial^2 U}{\partial r^2}} + \frac{1}{r} \frac{\partial U}{\partial r} - \frac{i\omega \rho}{\mu} U = \frac{G_0}{2\mu}.
$$

 $\left[\omega U \rho + \frac{V_0}{2} - \mu \frac{1}{r} \frac{\partial}{\partial r} (r \frac{\partial U}{\partial r}) \right] e^{i\omega t} = 0$ $\frac{z^{-2} \text{cos} \omega t}{\frac{\partial z^2}{\partial r^2} + \frac{1}{r} \frac{\partial z}{\partial r} - \frac{\omega_0}{r} \frac{\partial z}{\partial r}} = \frac{V_0}{2\mu}$.

Ill solution of $U(r)$ involves a complementary function, which is formu **Step 2** The full solution of $U(r)$ involves a complementary function, which is formulated with the Bessel function of the 1^st kind at 0th order, J_0 ; also the particular integral, $U_{pi} = -G_0/2i\omega\rho$:

$$
U(r) = \frac{iG_0}{2\omega\rho} \Big[1 - \frac{J_0(i^{3/2}\alpha \frac{r}{a})}{J_0(i^{3/2}\alpha)} \Big], \quad \text{with} \quad J_0(s) = \sum_{k=0}^{+\infty} \frac{(-1)^k}{k!k!} \left(\frac{s}{2}\right)^{2k},
$$

 ω \mathbf{v} .

and α denotes the non-dimensional **Wormersley number**: $\alpha = a \sqrt{\dfrac{\omega \rho}{\mu}}$ $\frac{d^2p}{\mu} = a \sqrt{}$

Step 3 To recover u_z from $U(r)$:

$$
u_z(r,t) = \frac{i}{\omega \rho} \frac{\partial p}{\partial z} \left[1 - \frac{J_0(i^{3/2} \alpha \frac{r}{a})}{J_0(i^{3/2} \alpha)} \right] = \frac{iG_0}{2\omega \rho} \left[1 - \frac{J_0(i^{3/2} \alpha \frac{r}{a})}{J_0(i^{3/2} \alpha)} \right] e^{i\omega t}
$$

Ostensibly, this solution is defined in the complex domain; but for simplicity, we only consider the real part to interpret its physical meaning.

Extended Properties

1. Wall shear stress:

$$
\tau_{rz} = \mu \frac{\partial u_z}{\partial r} = \mu \Re \left\{ - \frac{a}{i^{3/2} \alpha} \left(\frac{J_1(i^{3/2} \alpha)}{J_0(i^{3/2} \alpha)} \right) \frac{\partial p}{\partial z} \right\}, \quad \text{with} \quad J_n(s) = \sum_{k=0}^{+\infty} \frac{(-1)^k}{k!(k+n)!} \left(\frac{s}{2} \right)^{2k+n}.
$$

2. Volume flow rate:

$$
Q(t) = \int_0^a 2\pi r u_z \mathrm{d}r = \Re \Big\{ -\frac{\pi a^4}{i\mu \alpha^2} \Big(1 - \frac{2J_1(i^{3/2}\alpha)}{\alpha i^{3/2} J_0(i^{3/2}\alpha)} \Big) \frac{\partial p}{\partial z} \Big\}.
$$

The Wormersley Number The Wormersley number α is the ratio between unsteady inertia force and viscous force.

- $\cdot \alpha \leq 1$: **Quasi-steady**, the velocity profile is basically scaled Poiseuille flow, mainly observed in the microvasculatures (*e.g.*, capillaries, venules);
- \cdot α > 1: Oscillatory, the velocity profile is balanced between viscous forces at the wall and inertial forces in the centre. Common in large arteries (*e.g.*, ascending aorta, carotid artery).

FIG. 4: Womersley flow profiles. (a) Low α (viscous dominates), (b) intermediate α , (c) high α (inertia dominates).